

## PRELIMINALY STUDY OF THE INFLUENCE OF SOLAR CELL DEGRADATION DUE TO ESD ON SOLAR ARRAY POWER GENERATION

Teppei Okumura<sup>1</sup>, Kazuhiro Toyoda<sup>1</sup>, Shirou Kawakita<sup>2</sup>, Mitsuru Imaizumi<sup>2</sup>, Mengu Cho<sup>1</sup>

1. Department of Electrical Engineering

Kyushu Institute of Technology, 1-1 Sensui, Tobataku, Kitakyushu 804-8550, Japan

2. Japan Aerospace Exploration Agency

### ABSTRACT

In space, an electrostatic discharge (ESD) can occur on a solar array due to the plasma interaction. One of the issues of ESD is the degradation of solar cell electric performance. To establish the power degradation estimation method due to ESD in solar arrays, light current-voltage characteristics are evaluated in the current value at maximum power. From the results of the calculation, InGaP/GaAs/Ge solar arrays potentially suffer more serious power degradation than Si solar array.

### 1. Introduction

The electrostatic discharge, the so-called primary arc, occurs on solar arrays because of the plasma interaction. The detail phenomenon about primary arcs is described in Ref. 1. In order to establish an international standard for solar array primary arc tests, the collaboration research is carrying out in France, USA and Japan<sup>[2]</sup>. Because one of the issues of primary arcs is the degradation of solar cell electric performance<sup>[3][4]</sup>, the primary arc test on Silicon with integrated bypass function solar cell (Si w/IBF cell), Silicon solar cell (Si w/o IBF cell) and InGaP/GaAs/Ge solar cell (3J cell) have been carried out<sup>[5]</sup>.

Originally, only the radiation particles have been considered as the cause of solar cell degradation. Therefore, the solar cell degradation due to primary arcs should be combined with the estimation method for solar array power generation in satellite life. The solar array power degradation due to primary arcs is examined as a preliminary study in this paper.

### 2. Experiment

The 3J cell, Si w/IBF cell and Si w/o IBF cell are shown in Fig. 1. The solar cell performance is flight quality. The size of the cell is about 40mm×80mm. The solar cell is attached to an aluminum plate which was covered with a polyimide sheet using a silicone adhesive (RTV-S691). The thickness of the coverglass was 100μm. The bus bar was insulated with a silicone adhesive as in the real solar array design.

Fig.2 shows the light current-voltage characteristics (Light IV) and the light power-voltage characteristics (Light PV) of a 3J cell. Because the solar array is

operated under constant voltage, the current value at maximum power ( $I_{mp}$ ) is defined as the characteristic value of the light IV. The change in  $I_{mp}$  by one primary arc is defined as  $dI_{mp}$  obtained from eq.1.

$$dI_{mp} = \frac{(I_{mp\_before} - I_{mp\_after})}{I_{mp\_before} \times N_{arc}} \times 100, \% \quad (1)$$

The discharge circuit is shown in Fig.3. The primary arc experiment has been carried out under plasma conditions (LEO environment) and high energy electron beam conditions (GEO environment). The detail of the experiment condition is described in Ref.5.

A typical current waveform is a pulse whose duration is approximately 10μsec. We can adjust the peak value of the primary arc current and the primary arc energy by changing  $C_{ext}$ . The primary arc energy is obtained from two calculation methods. Under LEO environment, the primary arc energy ( $W_{arc\_LEO}$ ) is obtained from eq.2 where  $T_{ini}$  and  $T_{end}$  are the beginning and the end time of the primary arc current. Because of the malfunction of the voltage probe in a GEO environment, the arc energy ( $W_{arc\_GEO}$ ) is obtained from eq.3.  $Q_{arc}$  is the charge value of the primary arc<sup>[5]</sup>.

$$W_{arc\_LEO} = \int_{T_{ini}}^{T_{end}} I(t) \times V(t) dt, J \quad (2)$$

$$W_{arc\_GEO} = \frac{Q_{arc}^2}{2 \times C_{ext}}, J \quad (3)$$

### 3. Results and Discussion

Fig.4 shows the relationship between  $dI_{mp}$  and  $W_{arc}$  of a 3J cell.  $dI_{mp}$  increases with an increasing value of  $W_{arc}$ . The maximum  $dI_{mp}$  is 14% at 30mJ of  $W_{arc\_LEO}$ . The minimum  $dI_{mp}$  is 0.05% at 0.8mJ of  $W_{arc\_GEO}$ . Fig.5 shows the relationship between  $dI_{mp}$  and  $W_{arc}$  on a Si w/IBF cell and on a Si w/o IBF cell. The minimum  $W_{arc}$  for the degradation is 30mJ of  $W_{arc\_LEO}$ ;  $dI_{mp}$  is 0.2%. The maximum  $dI_{mp}$  is 1.5% at 280mJ of  $W_{arc\_GEO}$ . Because the silicon solar cell is more robust than the 3J cell as reported in ref.5,  $dI_{mp}$  of the Si cell is much smaller than that of the 3J cell under the same  $W_{arc}$ .

In general, the solar array is operated under a constant voltage. Therefore the output voltage of each solar cell is briefly determined by the number of solar cells and the output voltage of the solar array. In the case of the 3J solar array (output voltage; 60V, output current; 0.45A, output power; 27W, number of 3J cell;

30), the decrease of the output current by 14% in one 3J cell causes a decrease of 0.7% in output power.

**4. Summary and Future work**

In order to establish the estimation method for solar array power degradation due to primary arcs in solar arrays, the preliminary study was carried out. Because the solar array is operated under constant voltage, the light IV characteristic is evaluated at the current value at maximum power.

From the results of the examination, the decrease value of the current output is increasing with arc energy in the case of the 3J cell. In the worst case, the 3J cell loses around 10% of its output current from a single arc. In the case of the Silicon cell, the decrease value of the current output does not have a strong relationship with the arc energy. In the worst case, the Silicon cell loses around 1% of its output current from a single arc.

**Acknowledgement**

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**REFERENCES**

- [1] D, Hastings., H, Garrett.: Spacecraft-Environment Interactions, Cambridge University Press, 1st edition, 2000.
- [2] Cho, M., Goka, T.: Issues Associated with Standardization of Ground Test Methods of Electrostatic Discharge Phenomena on Spacecraft Surface, 56th International Astronautical Congress, Fukuoka, October 2005
- [3] K, Toyoda., T, Okumura., S, Hosoda., M, Cho.: Degradation of High Voltage Solar Array due to Arcing in LEO plasma Environment, 2005, Journal of Spacecraft and Rockets, Vol.42, No.5, pp.947-953.
- [4] T, Okumura., H, Masui., K, Toyoda., M, Cho.: Degradation of Electric performance due to Electrostatic Discharge on Silicon solar cell for Space, 2007, J. Japan Soc. Aero. Space Sci., Vol. 55, No.647, pp.590-596.
- [5] T, Okumura., S, Ninomiya., H, Masui., K, Toyoda., M, Imaizumi., M, Cho.: Solar cell degradation due to ESD for international standardization of solar array ESD test, 10th Spacecraft Charging Technology Conference, Biarritz France, June 2007.

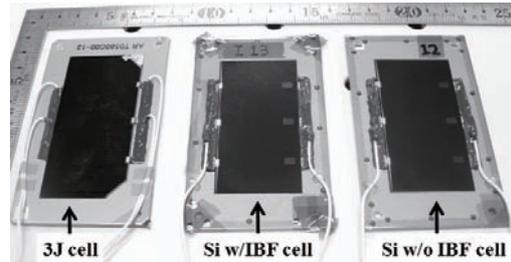


Fig.1 3J cell, Si w/IBF cell and Si w/o IBF cell

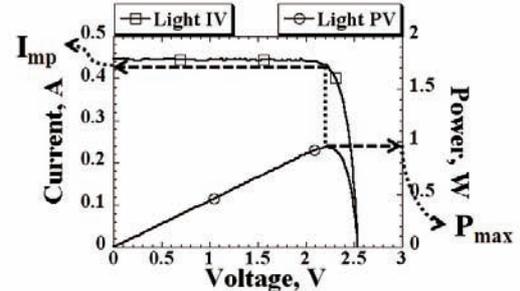


Fig.2 Light current-voltage characteristics and light power-voltage characteristics of a 3J cell

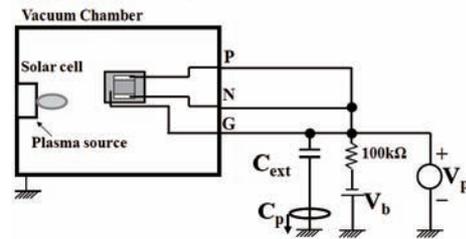


Fig.3 Discharge circuit

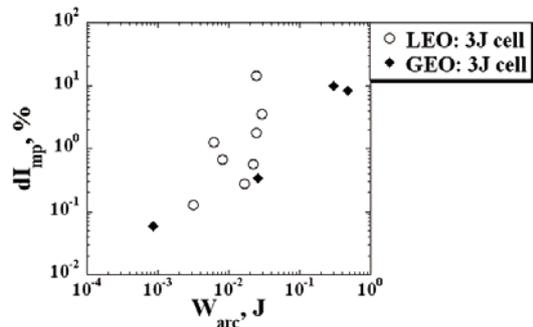


Fig.4 Relationship between  $dI_{imp}$  and  $W_{arc}$  on a 3J cell

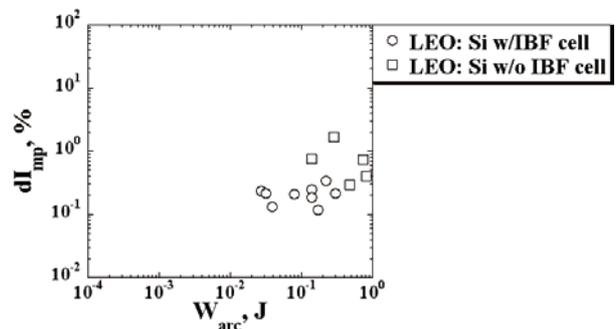


Fig.5 Relationship between  $dI_{imp}$  and  $W_{arc}$  on a Si w/IBF cell and on a Si w/o IBF cell