

アストロダイナミクスシンポジウム 2023.7.24

A robust method against Solar condition to generate Initial Template Image for Optical Navigation in Planetary Exploration

惑星探査での画像航法における太陽条件にロバストな初回マッチング用テンプレート作成手法

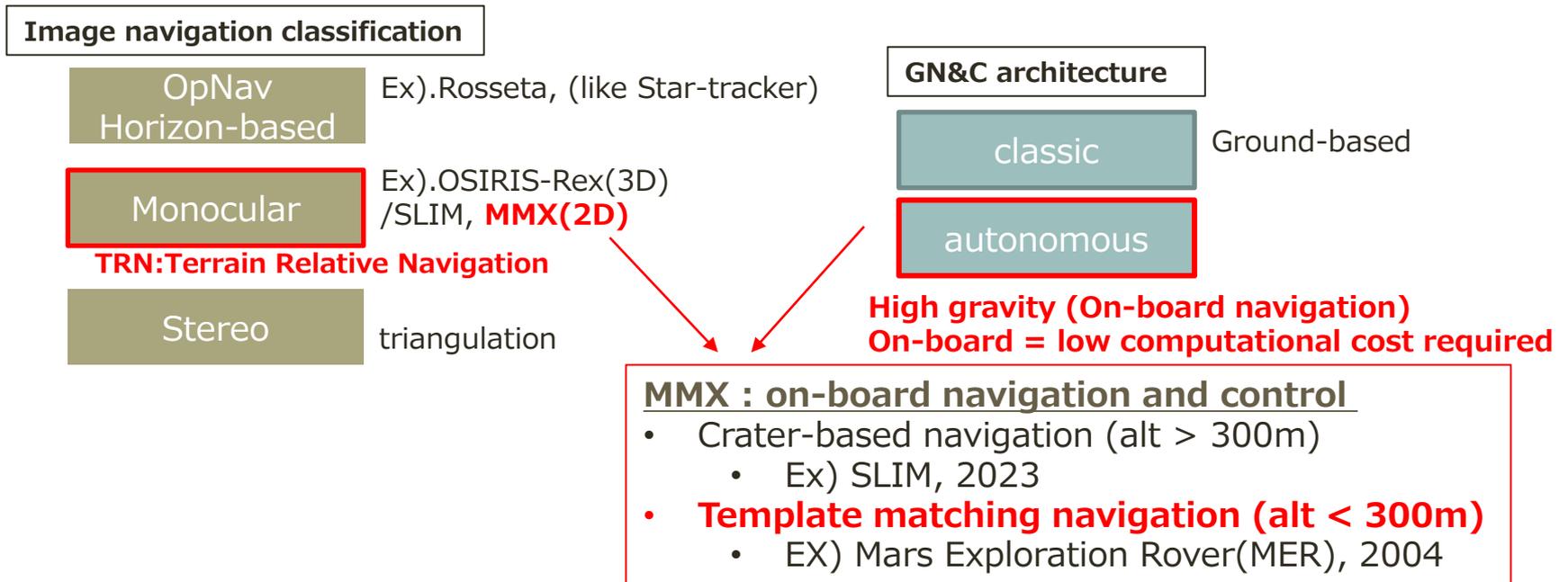
MMX
Martian Moons eXploration



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Background

- Image navigation for Small body landing in Planetary Exploration[1-5]

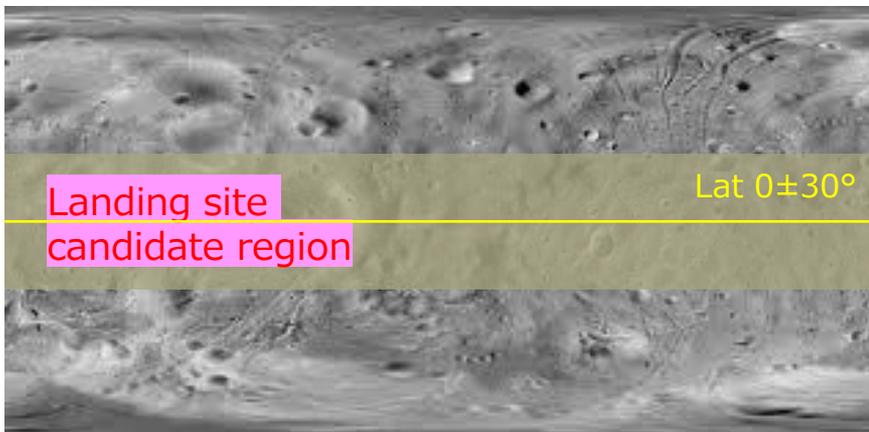


- Template matching is **primitive (classical)** method for image navigation
- Computational cost : **small(good)**
- Resource : **small(good)**
- Solar condition change : **weak**

In this study, we propose the robust method against solar condition

MMX mission

- MMX is **world's first sample return mission** from a Mars satellite[6][7]
- Decent operation
 - Rehearsal#1, Landing #1
 - Unti-Mars side
 - Rehearsal#2, Landing #2
 - Mars side (“blue” area)



(landing image)



- Launch : 2024
- Mass : About 4000kg
- Mission duration : About 5yr
- Mission Instruments
 - Sampler
 - Gamma-ray, neutron spectrometer
 - Near-infrared spectrometer
 - Wide FOV camera
 - Narrow FOV camera

Image navigation and decent sequence in MMX mission

- **Template matching** will be performed from **300m to 10m** altitude after crater matching navigation(CB).
- Due to residual error of CB, **error reset** is required by **initial template** which is uploaded from the ground, and **its source is captured image by CAM-T(telescopic)** in QSO-L.
- Templates of 2nd matching and later are generated by cropping from the previously taken image.
- The accuracy of **initial template matching** is very important because it directly affects landing accuracy.

Target Matching Navigation (Template matching)

Image	Camera
Target image	Wide FOV (CAM-W)
Initial template	Telescopic (CAM-T)

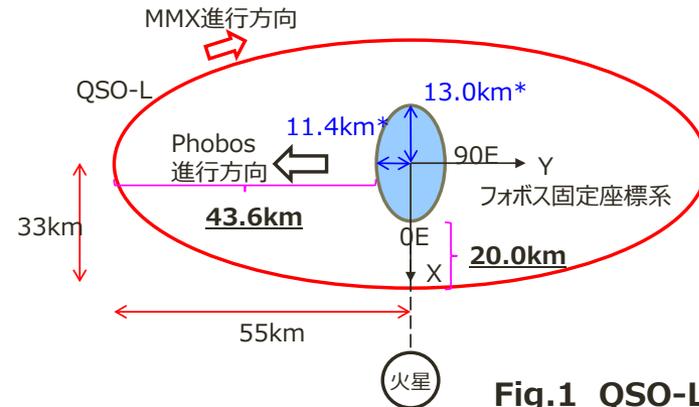


Fig.1 QSO-L[6]

*Willner et al., Phobos shape and topography models, 2014

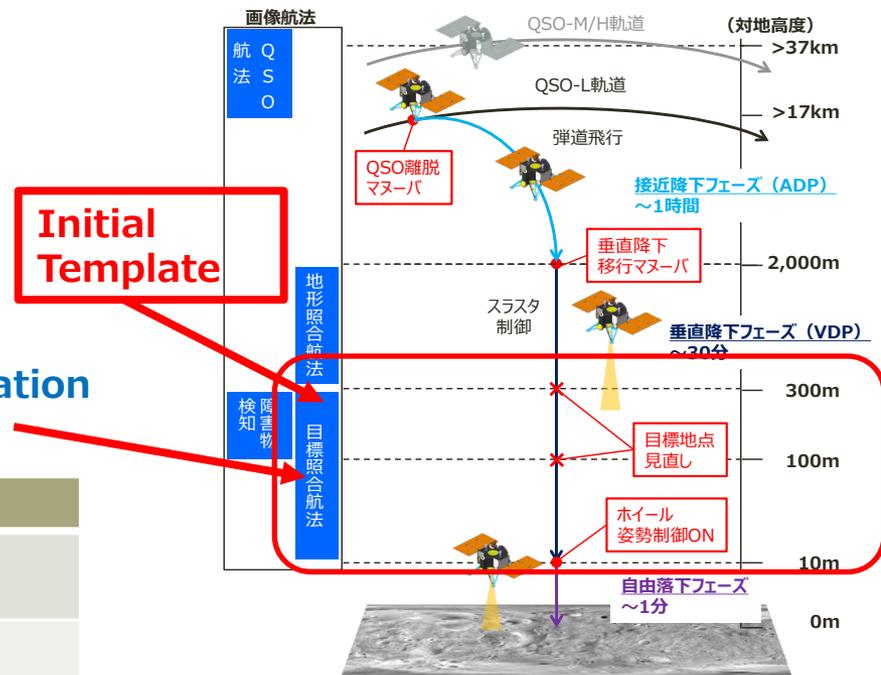
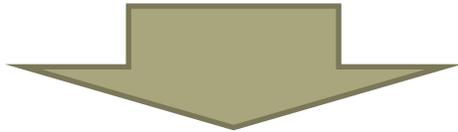


Fig.2 Image navigation and decent sequence[7]

Problem and proposed method

- Initial template matching is generated by using the image captured from QSO-L by CAM-T before decent operation.
- Decent operation will be performed at **daybreak**. However, **observation chances** of such a solar condition in QSO-L are **limited**.
- Thus, robust matching can facilitate observation operations in QSO.



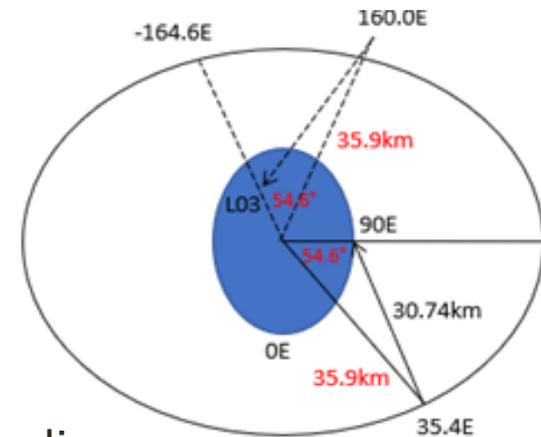
Proposed method(pre-processing):

- **Histogram equalization** for initial template to **enhance contrast** and to be **robust** for matching

Other operational innovations:

- Increase exposure time
- Oblique viewing to gain resolution in the case of far side (leading / trailing side)

Oblique viewing



Nadir

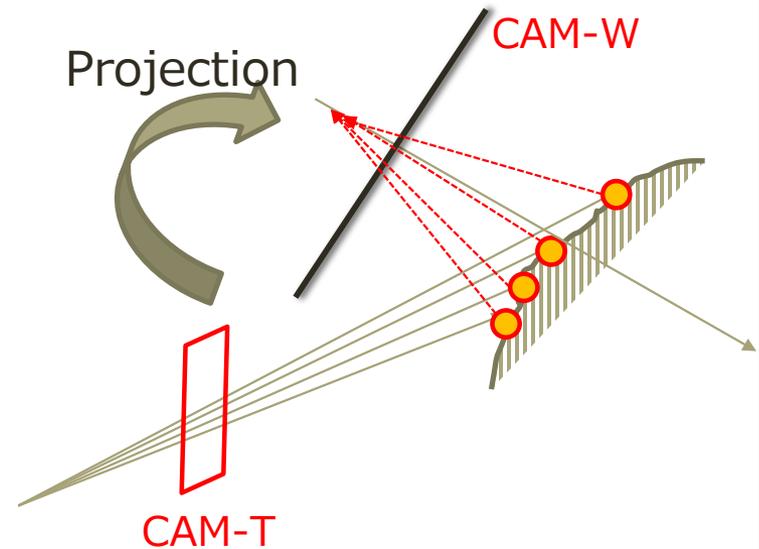


Procedure for generating template

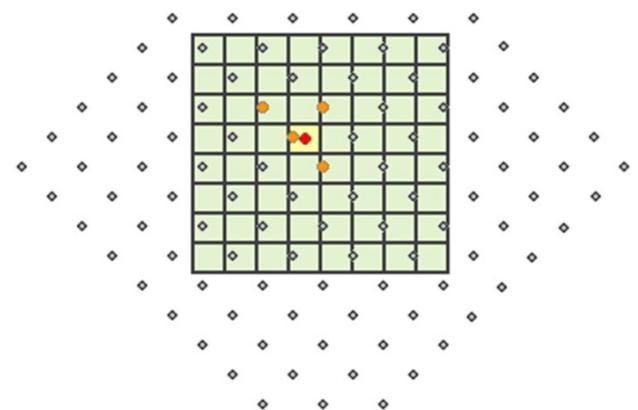
- Obtain **intersections with Phobos surface** by each pixel of **CAM-T** image captured from **QSO-L** using **DEM**
- **Project** intersections 3D coordinates with respect to Phobos fixed coordinate **onto CAM-W attitude and position** by following equation:

$$\begin{pmatrix} x \\ y \\ 1 \end{pmatrix} = P \begin{pmatrix} X \\ Y \\ Z \\ 1 \end{pmatrix}$$

- Where P is the perspective projection matrix, which can be calculated from $K[R \mid t]$, assuming K is internal camera matrix, R is rotational(attitude) matrix, and t is spacecraft position, respectively.
- Because the image coordinates x, y are the real-number, **resampling** is required (i.e., bi-linear interpolation)
- **Finally, pre-processing and trimming by template-size**



resampling



Template matching

- finding small parts which match a template image with maximum similarity

Sum of Absolute Difference(SAD)

$$S = \sum \sum |g(d_x + i, d_y + j) - f(i, j)|$$

In this study, utilize
SAD/NCC considering
computational cost

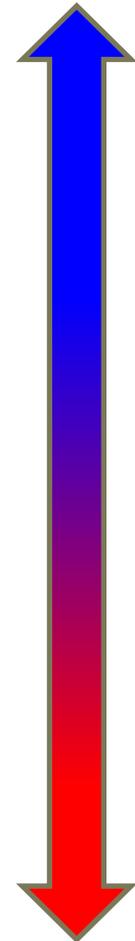
Normalized Cross-Correlation(NCC)

$$S = \frac{\sum \sum \{g(d_x + i, d_y + j) - f(i, j)\}}{\sqrt{\sum \sum (g(d_x + i, d_y + j))^2} \sqrt{\sum \sum (f(i, j))^2}}$$

Zero-means Normalized Cross-Correlation(ZNCC)

$$S = \frac{\sum \sum \{(g(d_x + i, d_y + j) - \mu_g) - (f(i, j) - \mu_f)\}}{\sqrt{\sum \sum (g(d_x + i, d_y + j) - \mu_g)^2} \sqrt{\sum \sum (f(i, j) - \mu_f)^2}}$$

Low Cost



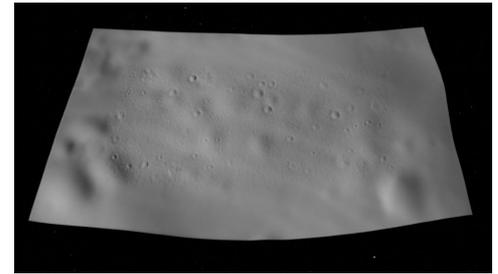
Robust

Simulation

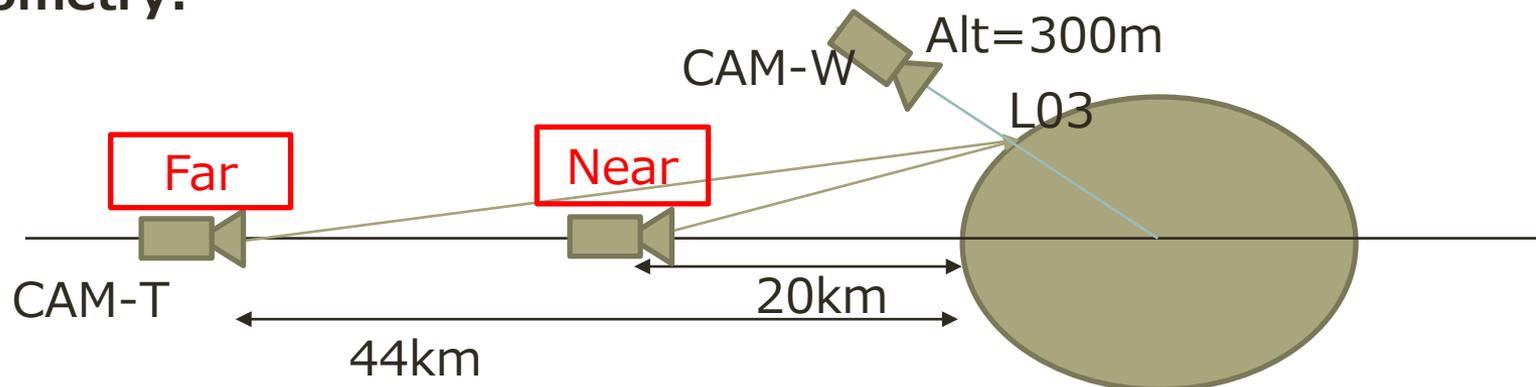
Simulation condition

- Template size (px) : 128 x 128
- Target image size (px) : 1024 x 1024
- Landing point : L03 (Candidate site)
- Solar conditions:
 - Target image
 - EL= 145°, AZ= 40° (FIX)
 - Template
 - EL, AZ = Variables
- Model : Hapke reflectance

L03 Landing candidate site
(Lat, Lon = 25.8°, -164.6°)

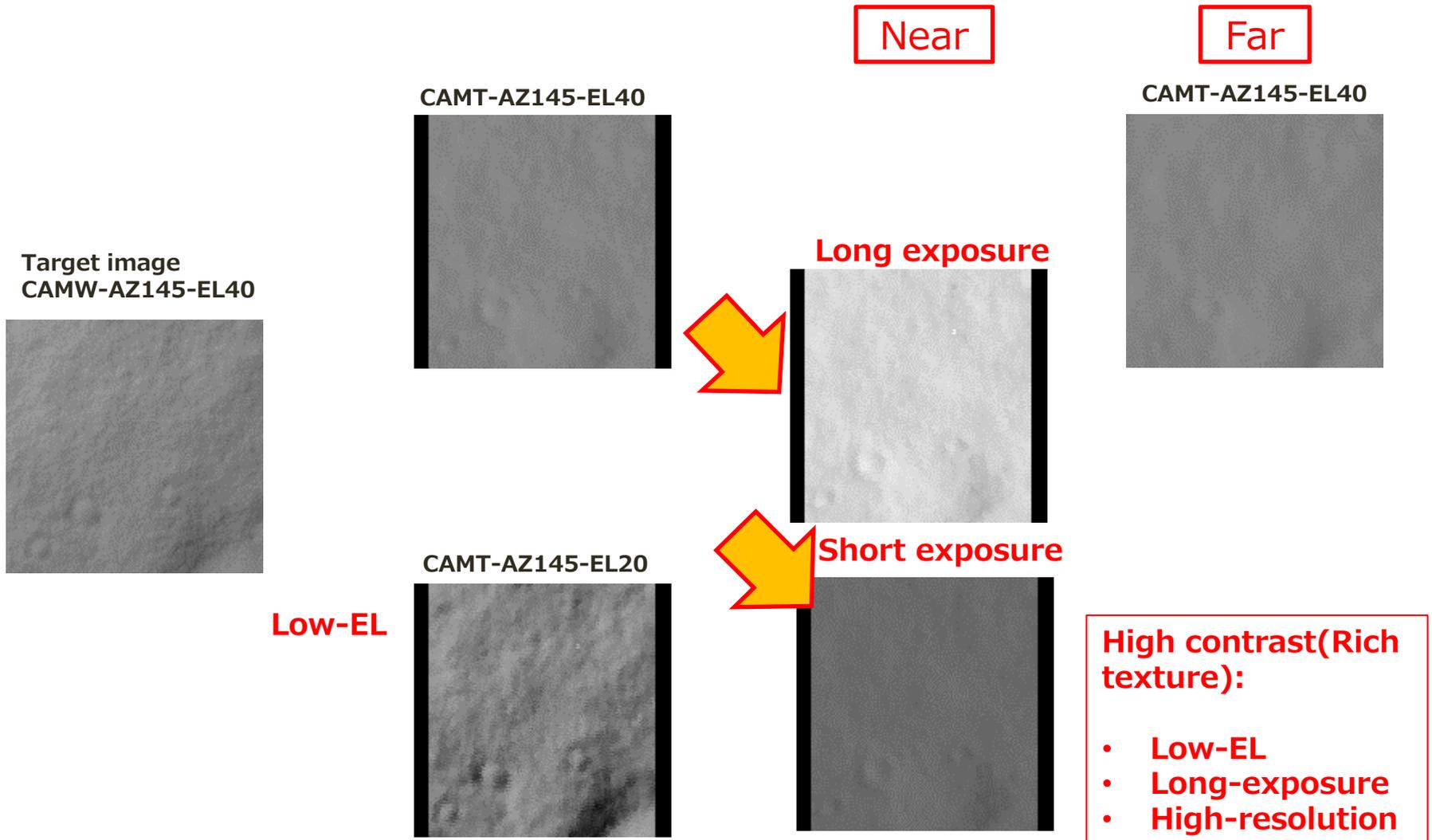


Geometry:



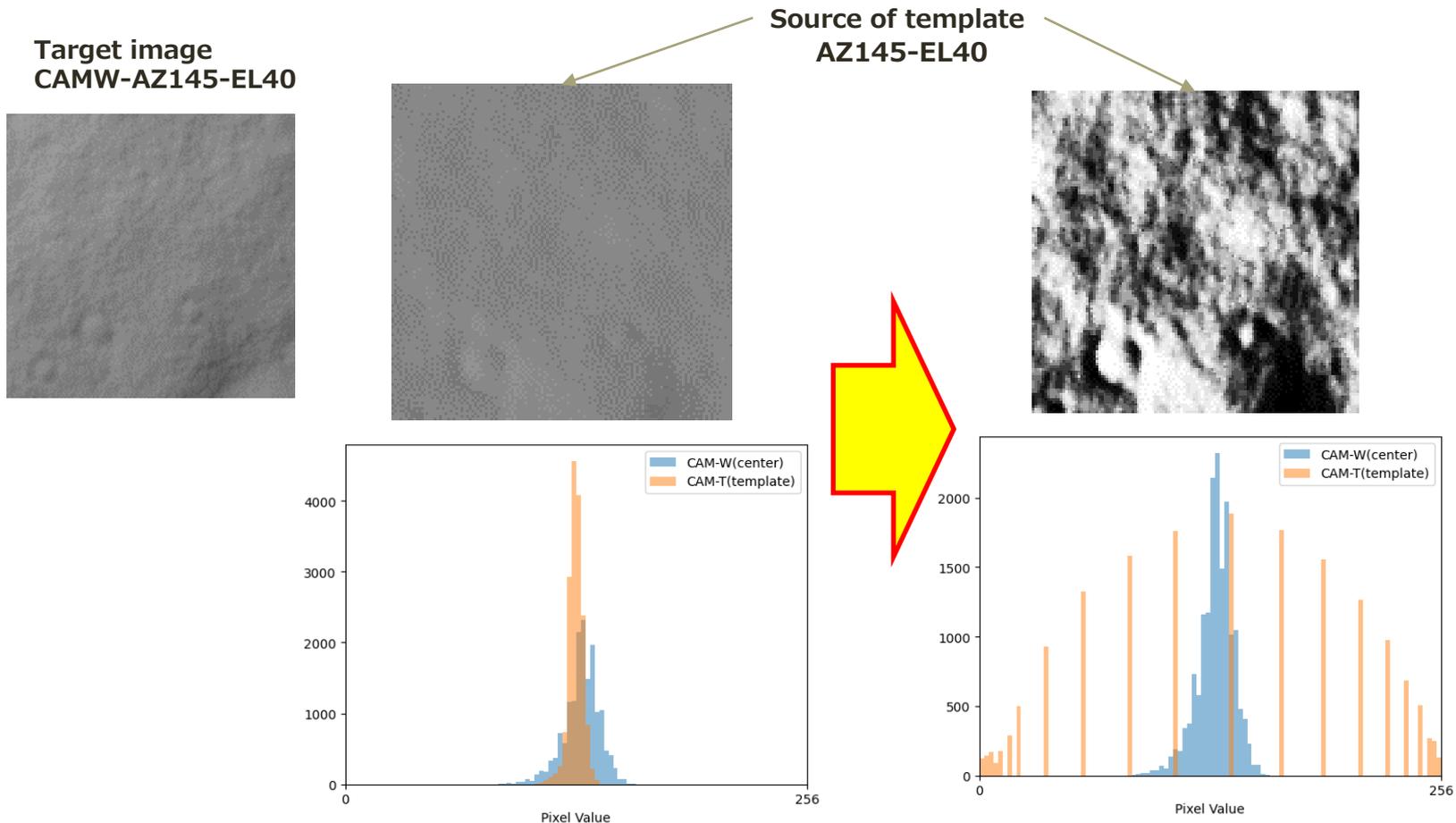
imageries

- Sources of template (after resampling)



Histogram equalization

- To enhance contrast of template, **histogram equalization** is applied to CAM-T image (cv2.equalizeHist)
- Average brightness should be set to **128**



Accuracy test against solar condition

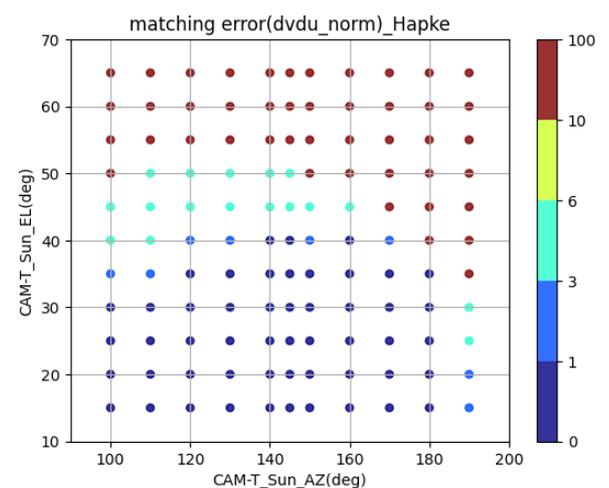
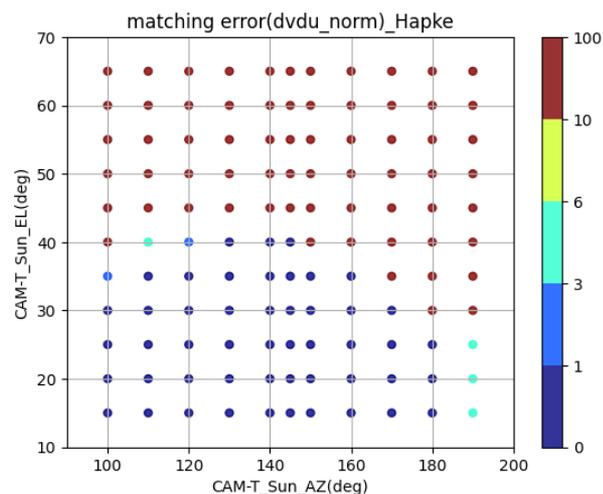
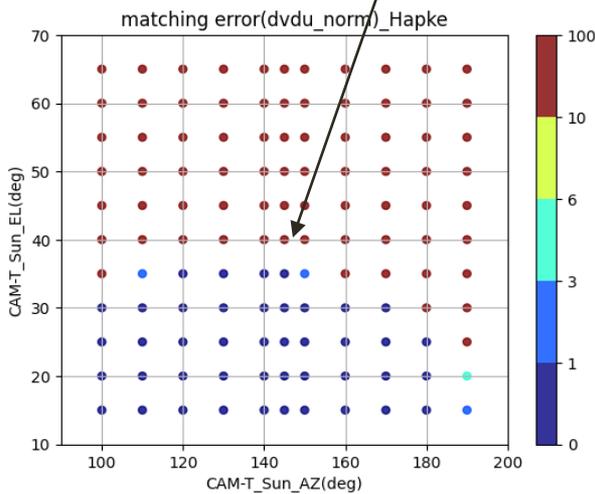
- Robustness can be increased by **histogram equalization**
- Matching algorithm = **NCC**

Near-side

CAM-W
AZ,EL

Long exposure

Histogram equalization

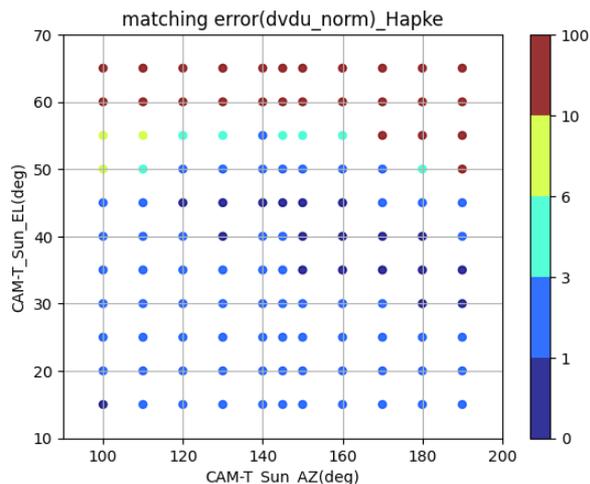
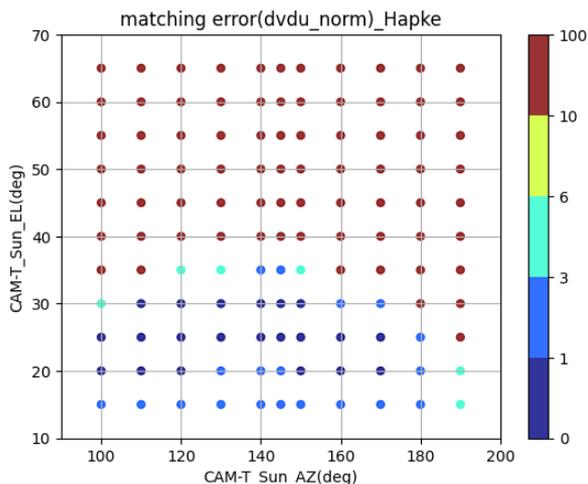


Accuracy test against solar condition

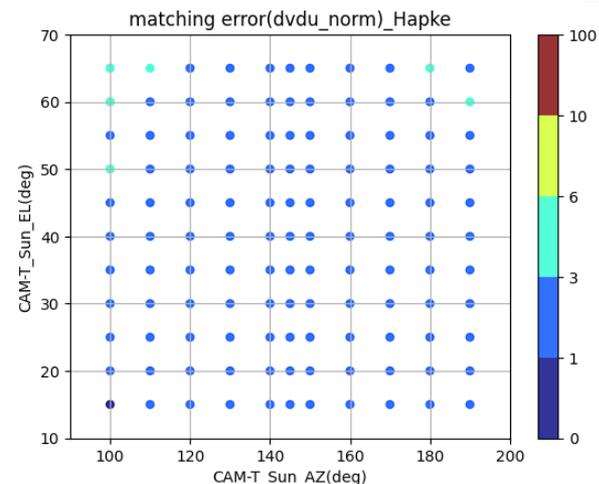
- Robustness can be increased by **histogram equalization and oblique viewing**
- Matching algorithm = **NCC**

Far-side

Histogram equalization



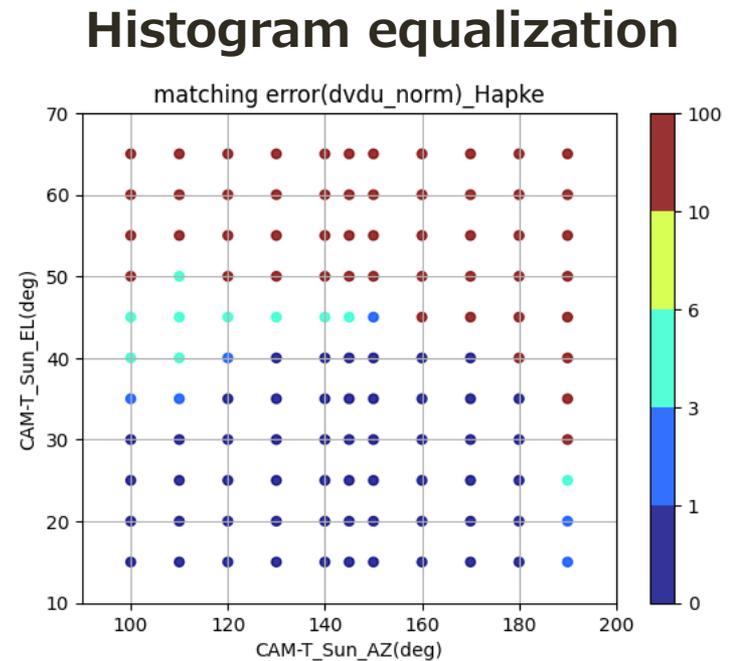
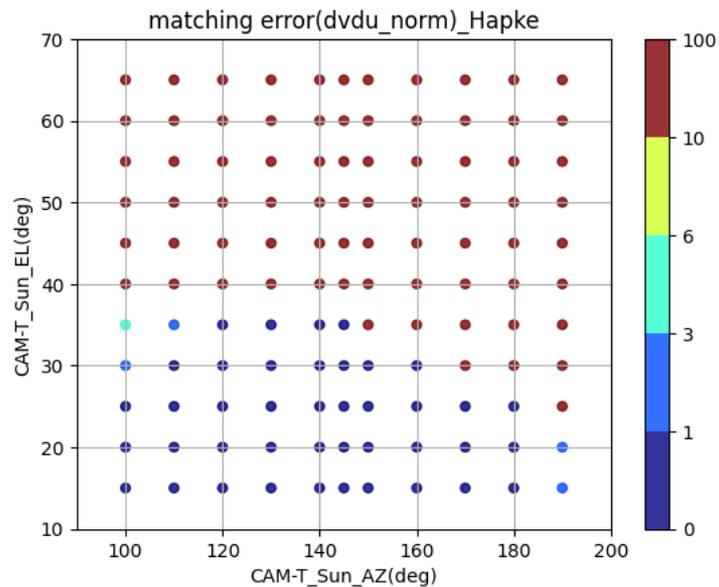
Oblique viewing + Histogram equalization



Accuracy test against solar condition

- Robustness can be increased by **histogram equalization**
- Matching algorithm = **SAD**

Near-side

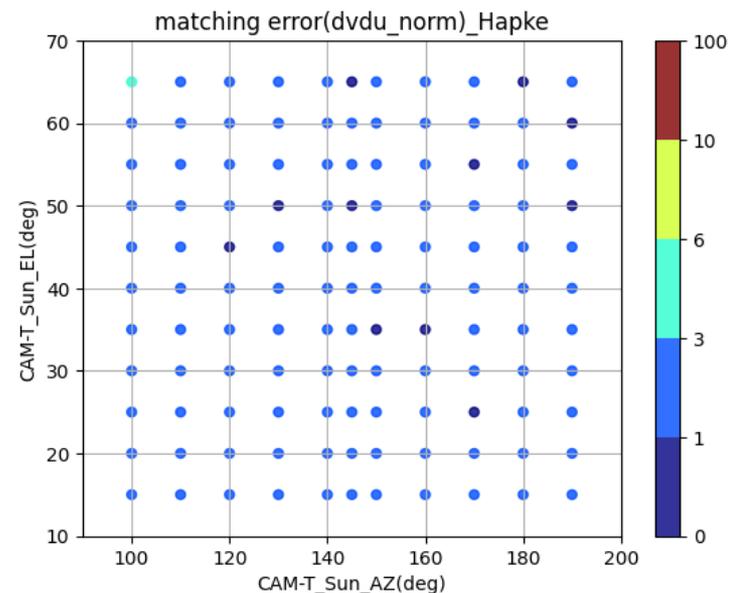
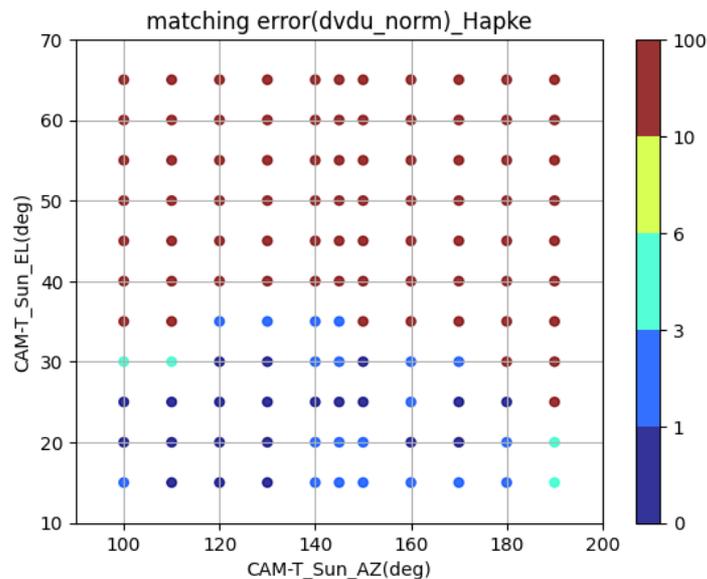


Accuracy test against solar condition

- Robustness can be increased by **histogram equalization and oblique viewing**
- Matching algorithm = **SAD**

Far-side

Oblique viewing + Histogram equalization



Conclusion & operation strategy

- **Initial template matching** accuracy and robustness is very important for safety landing of MMX.
- Pre-processing by **Histogram equalization** for template is effective for the robustness against solar condition, especially even when utilizing small cost algorithm **SAD/NCC**. It can keep accuracy and robustness regardless of small resource and cost navigation.
- Nevertheless, **lower sun elevation** than that of decent operation (CAM-W) is better (Around $EL=20-40^\circ$) .
- In the case of far-side, **oblique viewing** is required to gain resolution.

- **Applications of this study:**
 - Not only initial template matching but also other (after 2nd) matchings (if some templates can be uploaded considering spacecraft resource).
 - Not only small body landing mission but also other navigation mission (e.g., earth satellite, docking, debris, and so on) because template matching is simple algorithm and commonly used.

A satellite with solar panels and antennas is shown in orbit. The background is a view of Earth from space, with a dark, cratered lunar surface in the foreground. The satellite is the central focus, with its solar panels extended. The text is overlaid on the satellite.

ご清聴ありがとうございました
Thank you for your attention

References

- [1] High Performance Embedded Computing in Space: Evaluation of Platforms for Vision Based Navigation, George Lentaris, et al , Journal of Aerospace Information Systems, Vol . 15, No. 4, April 2018
- [2] A Survey of Vision-based Navigation Technology for Spacecraft, Ishida et al., JAXA-RM-19-005
- [3] ISAS News No.469
- [4] A NASA GN&C Viewpoint on On Board Processing Challenges to Support Optical Navigation & Other Autonomous GN&C Critical Functions, C.J.Dennehy, A.Wolf, European Workshop on On Board Data Processing (OBDP2019) 25-27 February 2019 ESA/ESTEC Noordwijk, The Netherlands
- [5] Ari Jonsson, Autonomy in Space Current Capabilities and Future Challenges AI Magazine Volume 28 Number 4 (2007)
- [6] 擬周回軌道上からの画像撮影による着陸目標点の地表面傾斜角推定, 岡田, 2022
- [7] 火星衛星探査計画MMX 航法誘導制御系の独立検証, 巳谷, 宇宙科学技術連合会, 2022