

H2

デブリ衝突損傷リスク解析ツールに関する最近の機能拡充

Latest Expansion of Tactical Utilities for Rapid ANALysis of Debris on Orbit Terrestrial

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デブリ衝突損傷リスク解析ツール(TURANDOT)は宇宙機設計支援ソフトウェアである。このソフトウェアを、より多くのユーザーが安心して利用できるツールとするために、過去2年間にわたって複数のベンチマークテストを実施した。また、直近ではユーザーの利便性を高めるために入力支援機能を大幅に強化中である。TURANDOT に関連する 2012 年度～2014 年度の活動状況を報告する。

The screenshot shows the 'Input Support for TURANDOT' window. It contains several sections for configuring simulation parameters:

- Division method setting:** A dropdown menu for 'Division method' is set to 'N intervals (given reference to Mars Equinox)', with a value of 'N = 10'.
- Time setting:** Two rows for 'Begin of analysis time' and 'End of analysis time'. Each row has fields for year, month, day, and hour. The 'Begin' row is set to 2009, 01, 01, 00. The 'End' row is set to 2013, 12, 31, 24.
- Target orbit (model 1):** A table of orbital parameters with columns for the parameter name, a value, a unit, and a power of r .

Parameter	Value	Unit	r Power
Semi-major axis [km]	7178.0	km	r^1
Eccentricity of the orbit	0.001	-	r^2
Orbit inclination [deg]	158.0	deg	r^3
Right ascension of the ascending node [deg]	120.0	deg	r^2
Argument of perigee [deg]	150.0	deg	r^2

A 'Division' button is located at the bottom right of the window.

TURANDOT

デブリ衝突損傷リスク解析ツールに関する 最近の機能拡充

Latest Expansion of Tactical Utilities for Rapid Analysis of Debris on Orbit Terrestrial

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6th Space Debris Workshop
@JAXA/Chofu, 2014/12/17~2014/12/19

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Contents

- Introduction
- What is TURANDOT
- Validation
- New function
- Demonstration

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Estimated Annual Collision

- 800km Altitude

Debris Size	0.1~1mm	1~10mm	1~10cm	10cm<
Collision [1/m ² /year]	100	0.01	0.0001	0.00001
Solution	<u>Protection</u>	?	???	Manuever

- Computation by MASTER© by ESA
- Ref. ISSN 1349-113X Proceedings of the 5th Space Debris Workshop

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Kinetic Energy @ Collision

- Is common object small enough?
- Ex. 3g/cm³, 1mm³, 16km/sec
 - $3\text{g} \times 0.1\text{cm} \times 0.1\text{cm} \times 0.1\text{cm} = 0.000003\text{kg}$
 - $0.5 \times 0.000003\text{kg} \times (16\text{km/sec})^2 = 768 \text{ J}$
- 800 J concentrate on 1mm², in <μsec.
 - Easy to penetrate spacecraft structure

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Kinetic Energy @ Collision

- Cube object, 3g/cm^3 , 16km/sec , 2015/1/1~2024/12/31
- Collision Frequency is Estimated by MASTER-2009©+TURANDOT (Normal to orbit direction)

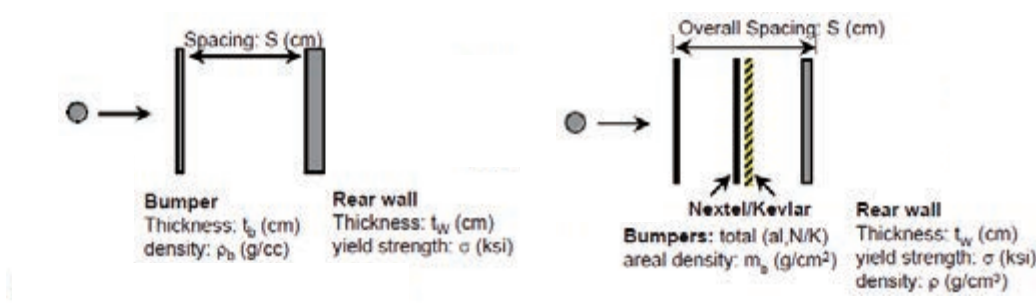
Debris Size [mm]	0.01~0.1	0.1~0.316	0.316~0.5	0.5~1
Collision Frequency [1/m ² /year]	646 (1/day)	67 (1/week)	0.6 (1/year)	0.13 (1/decade)
Kinetic Energy	<0.77J	<24J	<96J	<768J

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Protection Design

- Debris Bumper : Effective & Expensive



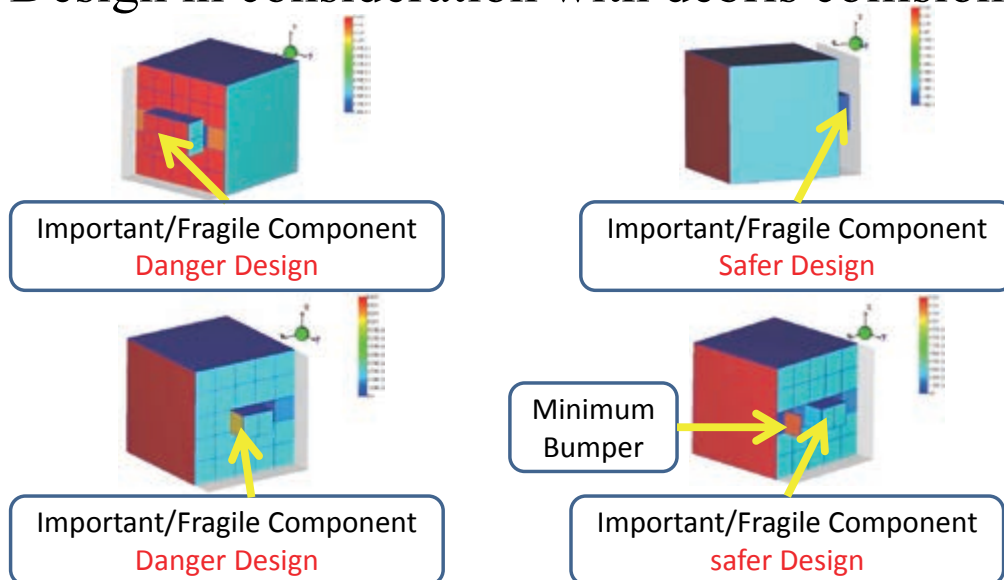
– Ref. IADC Protection Manual v5.0

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Protection Design

- Design in consideration with debris collision



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History

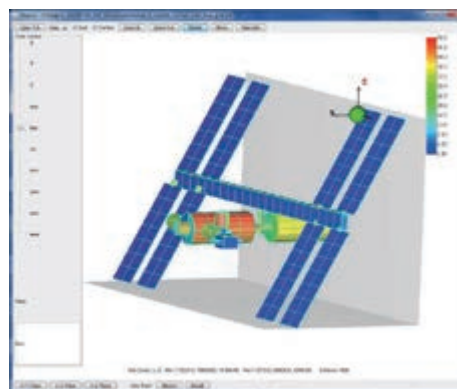
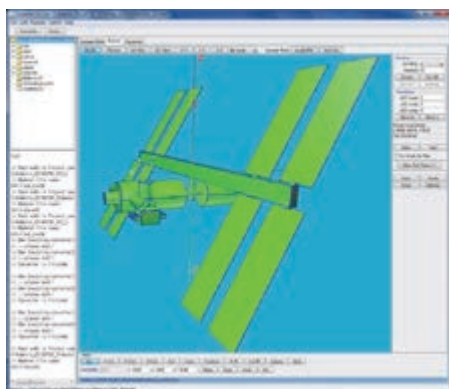
- April, 2007~Feb., 2009
 - “Collision Probability” Analysis Tool
- April, 2009~Feb., 2011
 - “Collisional Damage” Probability Analysis Tool
- April, 2011~Feb., 2012
 - Including “MASTER-2009”
- April, 2012~
 - Detail Modifications & Validations

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Functions I

- GUI
 - Integrated Analysis Environment
 - Satellite modeling
 - Grid generation



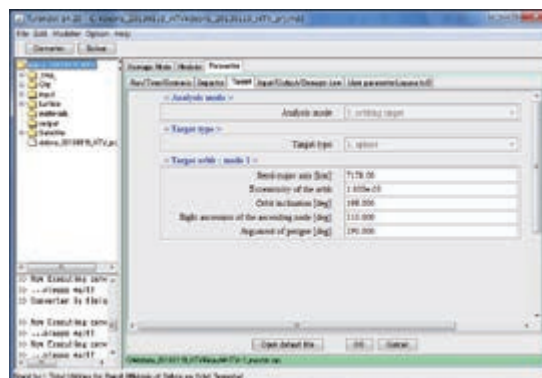
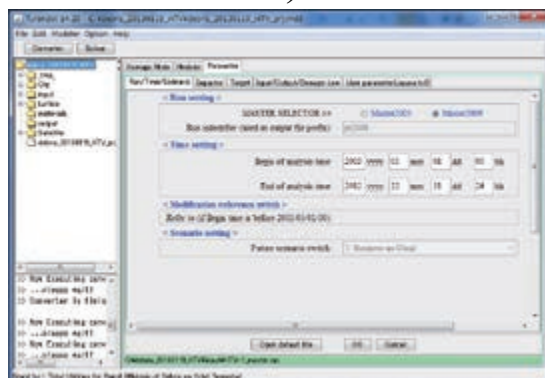
- Acknowledgement for Kurihara, M.

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Functions I

- GUI
 - Computation condition setting
 - Requirement from Databases (MASTER & ORDEM)



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Functions II

- No Domestic Database!
 - MASTER-2005 & ORDEM2000
 - MASTER-2009 & ORDEM2000

$$flux_{ORDEM} = flux_{MASTER} \cdot K \quad \text{if } 1 < K$$

$$flux_{ORDEM} = flux_{MASTER} \quad \text{if } K < 1$$

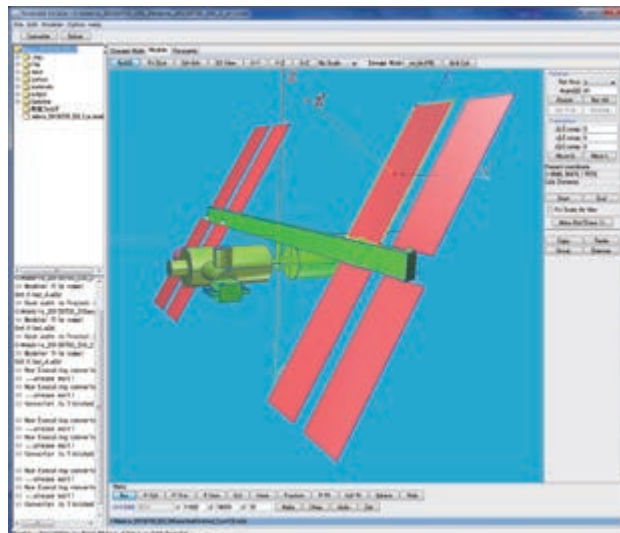
$$K = \frac{F_{ORDEM}}{\int_{4\pi} f_{MASTER} \cdot d\Omega}$$

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Functions III

- Damage Probability
 - Users' Definition
 - Fortran 95 like



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Result

Condition		TURANDOT	MASTER-2009	Error
a) $0.1\text{mm} < d$	Debris	6.3672	6.3678	0.0%
	Meteoroid	1.1050×10	1.1088×10	0.3%
b) $1\text{cm} < d$	Debris	1.5014×10^{-5}	1.5354×10^{-5}	2.2%
	Meteoroid	8.9217×10^{-7}	8.6510×10^{-7}	3.1%

- Good agreement
- TURANDOT is reasonable

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BLE Validation(1)

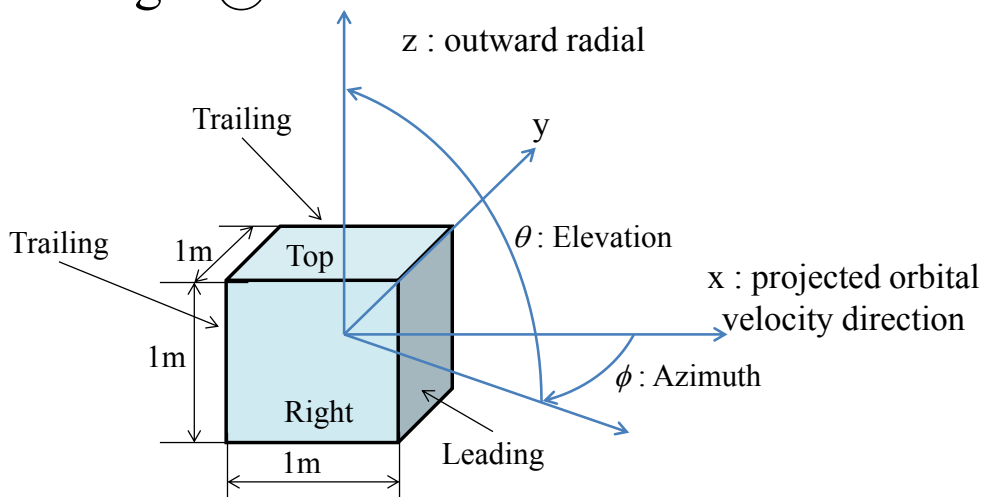
- MASTER-2005©
- Earth oriented attitude
- 2002/01/01~2002/12/31
- $0.001\text{m} < d < 0.1\text{m}$
- 85 debris flux data from “expl.cpe” as No.1001, 2001,..., 85001
 - 1 debris flux from every 1000 debris flux

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BLE Validation(2)

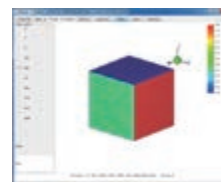
- Cube, (1m)³
- Damaged@1cm<d



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BLE Validation(3)



- Direction vector of debris flux

$$\mathbf{v}_{debris} = \begin{pmatrix} \cos \theta \cos \phi \\ -\cos \theta \sin \phi \\ \sin \theta \end{pmatrix}$$

- Normal vectors of the cube surfaces

Name	Normal Vector
Leading	(1,0,0)
Trailing	(-1,0,0)
Left	(0,1,0)
Right	(0,-1,0)
Top	(0,0,1)
Bottom	(0,0,-1)

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BLE Validation(4)

- Collision Probability for unit time

$$P_{\text{collision}} = \sum_{0 < \mathbf{V}_{\text{debri}} \cdot \mathbf{V}_n} \mathbf{V}_{\text{debri}} \cdot \mathbf{V}_n \times \text{flux}$$

- For Damage Probabillity, including ”condition”,

$$P_{\text{damage}} = \sum_{\substack{0 < \mathbf{V}_{\text{debri}} \cdot \mathbf{V}_n \\ \text{if } \text{contidion} = \text{true}}} \mathbf{V}_{\text{debri}} \cdot \mathbf{V}_n \times \text{flux}$$

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BLE Validation(5)

- Result

	MASTER2009 +MS-Excel©	TURANDOT
Leading	9.95E-09	9.95E-09
Trailing	1.21E-12	1.21E-12
Left	1.24E-09	1.24E-09
Right	5.62E-09	5.62E-09
Top	5.49E-12	5.49E-12
Bottom	3.04E-11	3.04E-11

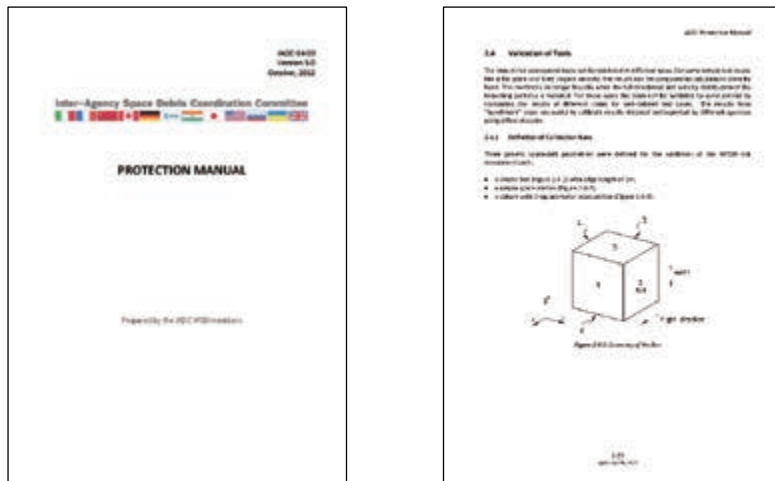
- Good agreement

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Validation

- IADC (Inter-Agency Space Debris Coordination Committee) Benchmark

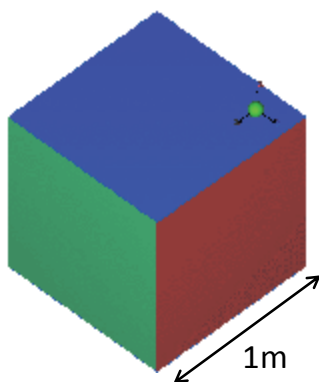


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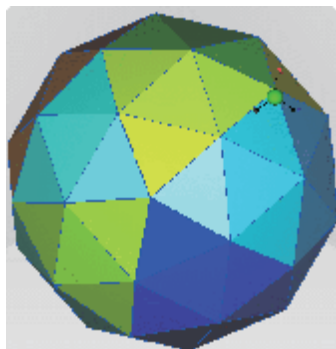
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IADC Benchmark

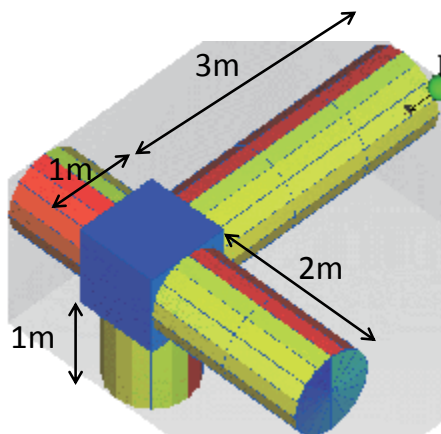
1) Cube



2) Sphere

Cross section 1m²

3) Station



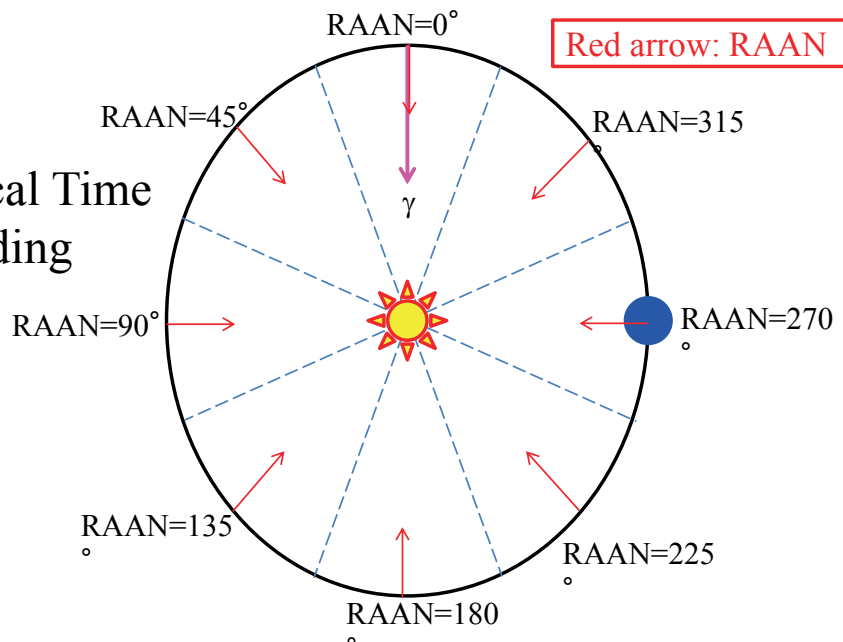
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Orbit Parameter Input Support (1)

- Example:

- SSO
- 12:00 Local Time on Ascending



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Result

- 2014/01/01~2014/12/31
- Solar Oriented, Cube of 1m³
- 1.E-5m < d

	W/O division	8 division	24 division
+X	108	110	105
-X	104	98	93
+Y	109	186	189
-Y	109	180	184
+Z	225	159	159
-Z	224	159	158

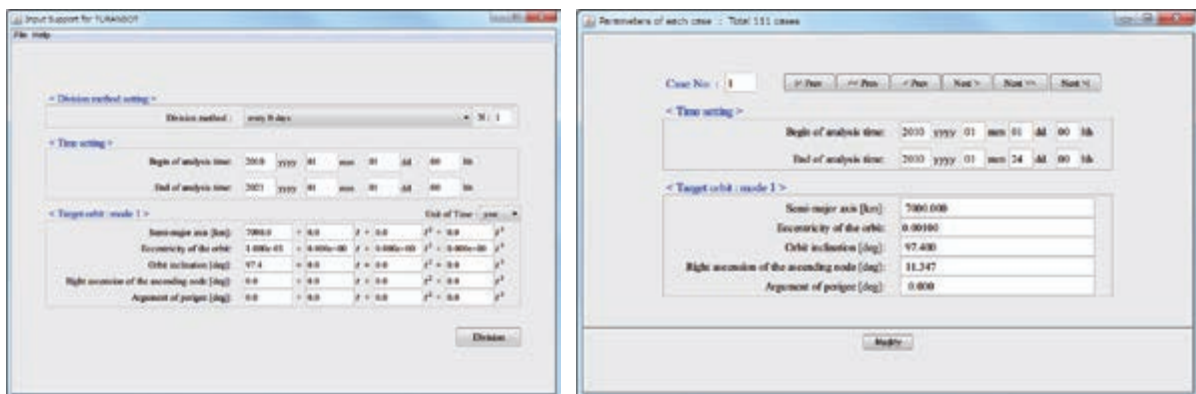
- Considerable difference

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Orbit Parameter Input Support (2)

- Automatic generation of input parameter



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Conclusion

- Validations and benchmarks are conducted.
- Modifications according to the results and users' opinion.
- New assistant functions.

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